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# INCREASE OF THE LOAD CARRYING CAPACITY OF ALUMINUM 2024-T3 BY MEANS OF A NiP-CrC-DLC COATING

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**Abstract:** The present investigation has been conducted in order to evaluate the tribological behavior of an AA2024-T3 aluminum alloy, coated with a NiP-CrC-DLC coating. The effect of NiP as intermediate layer was evaluated by carrying out calculations using ELASTICA® in order to determine its adequate thickness needed to avoid the plastic deformation of the substrate, ensuring then the integrity of the coating. To evaluate the efficiency of these calculations, a number of dry sliding wear tests were performed employing a ball-on-disk configuration, where alumina balls of 6 mm in diameter were used as counterpart. The sliding wear tests were carried out up to a sliding distance of 800 m, with a normal load of 5 N, a linear speed of 5 cm/s and a contact radius of 3 mm. The wear tracks were analyzed by means of scanning electron microscopy (SEM) techniques coupled with energy dispersive spectroscopy (EDS). The wear volume was determined by means of optical profilometry. The results indicate that, under the present testing conditions, the NiP-CrC-DLC coating exhibits a satisfactory behavior from the mechanical stability point of view when the thickness of the NiP layer is higher than 60  $\mu\text{m}$ , since no surface failures were observed at the end of the tests. For the coated system, the magnitude of the friction coefficient was found to be of approximately 0.1 and that of the wear rate was of about  $2.31 \pm 0.09 \times 10^{-16} \text{ m}^3/\text{N.m}$ . On the contrary, for the uncoated substrate, the friction coefficient was of approximately 0.5 and the wear rate of  $5.46 \times 10^{-13} \text{ m}^3/\text{N.m}$ , that is to say, 3 orders of magnitude greater than that determined for the coated system.

**Keywords:** AA2024-T3 aluminum alloy, duplex coatings, NiP, DLC, nanostructured coatings, PAPVD, sliding wear.

## 1. INTRODUCTION

Aluminum and its alloys exhibit excellent properties such as very high specific strength (strength/density), high toughness, ease of manufacture (major advantage being extrusion), long term performance, no low temperature brittle fracture, readily joinable by welding or mechanical riveting and easily recyclable. All these properties have conferred them an important place traditionally in the aerospace and automobile industries and lately, even in the fabrication of medical devices, where aluminum alloys may have a significant role to play with the potential to reduce manufacturing costs and to give ergonomic advantages to the surgeon due to a reduction in instrument mass [1].

However, the aluminum alloys show poor tribological resistance, as well as corrosion related problems such as localized corrosion, which limit their use for components that have to withstand either contact conditions or harsh environments or both. In order to overcome these drawbacks, extensive research was carried out since their development and different surface engineering methods were considered such as anodizing, heat treatment, laser modification, cladding, PVD and CVD techniques among others, able to improve their performance. Sometimes, it is found that the benefits gained in terms of corrosion protection could be offset by detrimental effects on fatigue performance or wear and vice versa [2].

One of the materials that have gained utmost importance as an attractive candidate for surface protection of these light alloys is diamond-like carbon (DLC) coating due to its high hardness, low friction nature, high chemical stability and excellent corrosion resistance. Unfortunately, when this type of coating is deposited on light alloys, the large Young's modulus mismatch between coating and substrate coupled with the existence of residual stresses, resulted in a poor load carrying support, which produces the cracking of the coating that is unable to follow the plastic deformation of the substrate.

As consequence, the load-carrying capacity related to the ability of the coating/substrate system to withstand the normal and tangential forces created by the system in a way that the functional properties of the coated system will remain unchanged, has constituted a main concern from the pioneering research carried out by Bull [3], related to this type of coatings, up to the present days.

The application of hard coatings on aluminum alloys that will permit the endurance of much higher contact stresses than would normally be the case of a bare alloy, requires either the use of different interlayers which act as load-supporting such as an micro-arc discharge oxidation alumina layer [4], a HVOF WC-CoCr thermal spray coating [5], an electrolytic Ni plating coating [6], the use of multilayers [7, 8] or gradient layers [9, 10], capable to decrease gradually the existing differences in properties between substrate and coating, amongst others.

Therefore, the present investigation has been conducted in order to evaluate the tribological performance of a system where an intermediate layer of electroless nickel has been used between the DLC (a-C:H)/CrC coating top layer and the AA2024-T3 substrate, in order to provide a sufficient load carrying capacity. Both the electroless NiP and DLC coatings were deposited industrially.

## **2. MATERIALS AND EXPERIMENTS**

### **2.1 Determination of the minimal thickness of the intermediate NiP layer in order to avoid plastic deformation of the substrate.**

Prior to the sample deposition, a number of calculations were carried out in order to determine the adequate intermediate NiP layer thickness needed to avoid the plastic deformation of the substrate. Such calculations were conducted employing ELASTICA®, a commercial software, which uses an extended Hertzian theory to solve elastic contact problems of layered structures. In this way, it is possible to ensure the integrity of the DLC coating, since the failure of such films depends strongly on the mechanical properties of the substrate.

It is well known that for the contact plane-sphere, at the beginning of the wear test, the developed stresses present their maximum values, which results in the maximum deformation of the substrate.

The elastic properties of both DLC (known industrially as Dymon-iC™, which is a hydrogenated amorphous DLC coating-a-C:H) and the NiP electroless coating, used in the present calculations, have been already determined and published elsewhere [11-13] (see Table 1). However, some of the mechanical properties shown in this Table were found from the existing literature on this topic and are indicated accordingly.

In the calculations, for the purpose of simplification, the thickness of DLC + CrC coatings was taken as 2.2 μm, considering only one film with similar properties, since the chromium carbide is likely to contain some H, although not as much as the top layer of the a-C:H coating. It has also been reported that the top layer coating could have approximately 30-40 at% H, with a small residual content of Cr, as consequence of the deposition procedure employed [13].

The mechanical behavior of the adopted duplex structure was evaluated by the analytical calculation of the elastic contact deformation fields under a spherical indenter for the cases of Hertzian contact conditions. The contact conditions: an alumina spherical indenter of 3000 μm radius (Preciball USA Inc.) and 5 N applied normal load, were also used during the tribological characterization, since it was stated that for the load-carrying capacity evaluation of the coating/substrate systems, the ball-on-disc test configuration provided the most relevant information [16].

Table 1. Characteristics and mechanical properties of the system under study

Material	Thickness ( $\mu\text{m}$ )	E (GPa)	HV (GPa)	$\nu$	$\sigma_y$ (GPa)
Al2024-T3 alloy	infinite	71.8 <sup>[14]</sup>	1.50	0.33 <sup>[14]</sup>	0.345 <sup>[14]</sup>
NiP electroless coating	To be determined	160 <sup>[11]</sup>	7 <sup>[11]</sup>	0.30 <sup>[11]</sup>	2.30 <sup>(14)</sup>
Alumina ball	-	360	17	0.20	23 <sup>[15]</sup>
Dymon-iC <sup>TM</sup>	2.2 <sup>[12]</sup>	146 <sup>[13]</sup>	18 <sup>[13]</sup>	0.3	10 <sup>[15]</sup>

The results obtained from ELASTICA© are shown in Figure 1, where two system architectures were considered: alumina ball/AA2024-T3 and alumina ball/DLC/NiP/AA2024-T3. The graph represents the variation of the calculated von Mises stress distribution for both configurations versus depth, for different NiP interlayers thicknesses. The value of the aluminum yield strength is also represented on the graph.

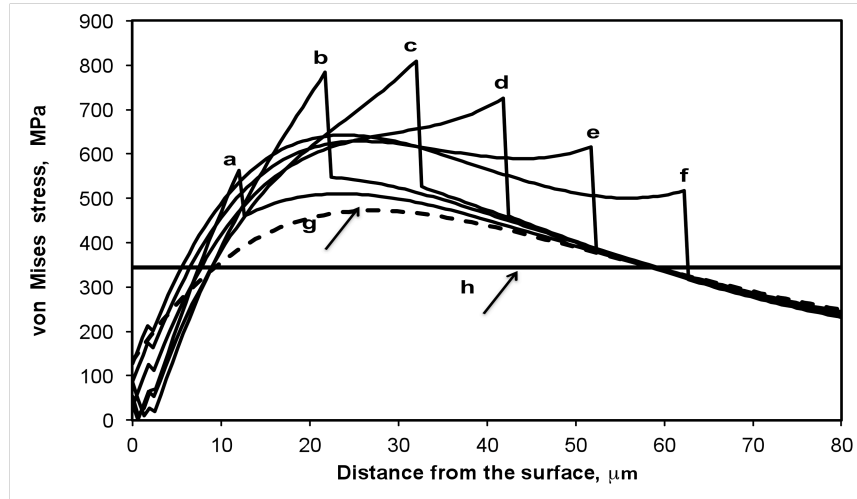


Figure 1. Variation of the von Mises stress with depth for both system architectures under study. **a:**  $\sigma_{\max} = 560$  MPa at  $d = 12$   $\mu\text{m}$  for NiP interlayer = 10  $\mu\text{m}$ ; **b:**  $\sigma_{\max} = 780$  MPa at  $d = 22$   $\mu\text{m}$  for NiP interlayer = 20  $\mu\text{m}$ ; **c:**  $\sigma_{\max} = 810$  MPa at  $d = 32$   $\mu\text{m}$  for NiP interlayer = 30  $\mu\text{m}$ ; **d:**  $\sigma_{\max} = 730$  MPa at  $d = 42$   $\mu\text{m}$  for NiP interlayer = 40  $\mu\text{m}$ ; **e:**  $\sigma_{\max} = 630$  MPa at  $d = 28$   $\mu\text{m}$  for NiP interlayer = 50  $\mu\text{m}$ ; **f:**  $\sigma_{\max} = 645$  MPa at  $d = 24$   $\mu\text{m}$  for NiP interlayer = 60  $\mu\text{m}$ ; **g:**  $\sigma_{\max} = 480$  MPa at  $d = 27.13$   $\mu\text{m}$  for the uncoated AA 2024-T3 substrate; **h:** yield stress of AA 2024-T3 substrate = 345 MPa.

It can be clearly seen that for the NiP interlayer of 60  $\mu\text{m}$  in thickness, the maximum Von Mises stress of 645 MPa takes place at a depth of approximately 24  $\mu\text{m}$ , i.e. inside the NiP coating, without exceeding its yield stress value of 2300 MPa. Moreover, from the calculations under Hertzian (only normal load) contact conditions it was determined that the contact radius for this system is of 46.3  $\mu\text{m}$  and the principal stress  $\sigma_1 = 1$  GPa for  $x=0$ , as compared to 56  $\mu\text{m}$  and  $\sigma_1 = 0.65$  GPa, respectively, the later corresponding to the alumina ball/uncoated AA2024-T3 substrate system. This increase in the compressive stress was attributed to the increase in the contact stiffness, as consequence of the introduction of NiP intermediate layer coupled with the redistribution of the surface contact stresses. A similar increase has also been reported by Bemporad *et al.* [17] for the PVD/HVOF duplex coating deposited onto a Ti6Al4V substrate

Furthermore, it is noticed that at a depth of 60  $\mu\text{m}$ , corresponding to the NiP intermediate layer, and for a normal load of 5 N, the von Mises stress value is less than 0.345 GPa, which corresponds to the value of the yield stress of AA 2024-T3 substrate, indicating that no plastic deformation of the latter takes place. Taking these calculations into account, it was decided to consider the deposition of the electroless NiP onto the aluminum alloy substrate, as an intermediate layer, followed by a thin DLC based top layer.

## 2.2 Coated system

Samples of an aluminum 2024-T3 alloy were mirror polished ( $R_a = 0,04 \pm 0,01\mu\text{m}$  by means of standard metallographic procedures. The NiP electroless coating with 11%P and a thickness of  $65 \pm 1\mu\text{m}$  and  $R_a = 0,20 \pm 0,01\mu\text{m}$  was deposited by Reliable Plating Inc. (USA). Subsequently, a DLC hydrogenated coating, commercially known as Dymon-iC TM (hydrogenated a-C:H) was deposited by Teer Coatings, England, by means of closed field unbalanced magnetron sputtering ion plating (CFUBMSIP), coupled with plasma assisted chemical vapor deposition (PACVD). In order to ensure a better adhesion of the DLC coating, a Cr film was deposited on top of the electroless NiP, followed by a coating of CrC and finally the a-C:H film. The sample preparation prior to deposition and the complete description of the procedure have been reported elsewhere [12, 13].

## 2.3 Tribological tests

Sliding wear tests were carried out in air ( $55 \pm 5\%$  humidity) up to a sliding distance of 800 m, with a normal load of 5 N, a linear speed of 5 cm/s and a contact radius of 3 mm, employing an alumina ball of 6 mm diameter, according to the ASTM G99-95. All the reported results represent mean values of at least 3 test runs. The wear tracks were analyzed by means of scanning electron microscopy (SEM) techniques coupled with energy dispersive spectroscopy (EDS). The wear volume was determined by means of a Zygo optical profilometer, which allowed the calculation of the wear constant,  $k$  ( $\text{m}^3/\text{Nm}$ ).

## 3. RESULTS AND DISCUSSION

### 3.1. Friction coefficients and wear volume

The variation of the friction coefficients with the sliding distance for both systems under study are presented in Figure 2. As it was expected, the friction coefficient for the alumina/AA2024-T3 alloy has a higher value as consequence of the adhesion phenomena, which take place between the harder alumina pin and the alloy. In this case, the stationary friction coefficient value was achieved at around 150 m, with an average value of 0.5. During the test, the adhered alloy substrate undergoes strain hardening and subsequently fatigue, as a consequence of the cyclic nature of the test, allowing therefore the formation of harder debris, which remain in the contact. These are embedded on the aluminum alloy, giving rise to a new surface that is much harder than the former one.

For the alumina/NiP/DLC system, values of the steady friction coefficient of around 0.1 were obtained. Considering the nature of the counterpart and the presence of the humidity approximately 60%, such values are similar as those reported by other authors (only for the systems for which the coating maintained its integrity without fracturing) [18, 19].

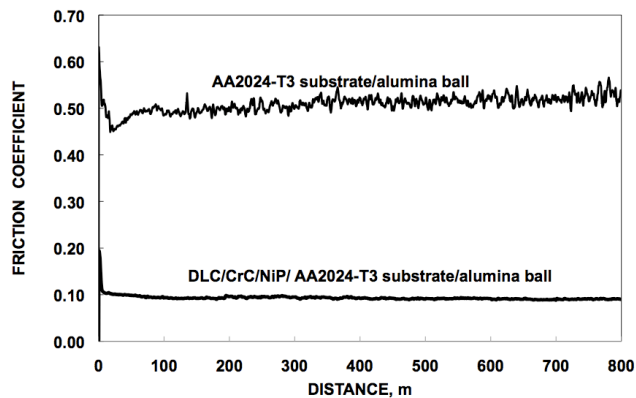


Figure 2 Evolution of the friction coefficient with the sliding distance.

The characteristic profiles of the wear tracks after the sliding tests are presented in Figure 3. The calculation of the wear volume was carried out by determining the area of the worn track from the 2D profile and further multiplying it by the circular periphery corresponding to a 3 mm radius. The wear rate,  $k$ , was calculated as the volume loss per unit of applied load and unit of sliding distance. The values are presented in Table 2.

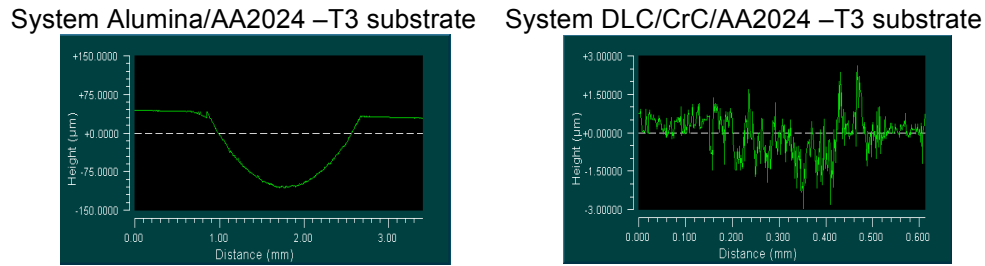


Figure 3. Characteristic profiles of the wear tracks.  
(Normal load=5 N, sliding velocity=0.05 m/s and  $4.3 \times 10^7$  cycles)

Table 2. Results from the tribological tests.

System	Load (N)	Distance (m)	Area (mm <sup>2</sup> )	Volume (mm <sup>3</sup> )	$k \times 10^{-16}$ (m <sup>3</sup> /N.m)
Alumina ball/AA2024-T3	5	800	$0.115 \pm 0.014$	$2.185 \pm 0.271$	$5462 \pm 0.03$
Alumina ball/DLC-CrN-NiP/AA2024-T3			$0.000048 \pm 0.000018$	$0.0009 \pm 0.0003$	$2.31 \pm 0.09$

As it can be observed, the wear constant value,  $k$ , for the uncoated aluminum alloy against the counterpart of  $5.46 \pm 0.00003 \times 10^{-13} \text{ m}^3/\text{N.m}$  is approximately 2400 times greater than that for the coated system, which is equal to  $2.31 \pm 0.00003 \times 10^{-16} \text{ m}^3/\text{N.m}$ . These values have the same order of magnitude as those reported by Stallard *et al.* [13], who obtained a sliding wear rate of  $5.1 \times 10^{-16} \text{ m}^3/\text{N.m}$  for the same DLC coating against WC balls as counterpart.

However, it is interesting to compare these values with those reported by Bolleli *et al.* [5], who carried out tribological tests under similar contact conditions (alumina ball of 6 mm and 5 N normal load) of DLC coatings deposited onto an AA6068-T6 substrate, using a WC-CoCr thermal sprayed coating of 125 μm in thickness as intermediate layer. The number of cycles corresponding to the wear test was of  $10 \times 10^7$ , at a sliding velocity of 0.3 m/s, which is 6 times higher than the sliding velocity employed in the present work. Nevertheless, the friction coefficient reported was of 0.1 and the wear rate was of order of  $10^{-16} \text{ m}^3/\text{N.m}$ , values which are consistent with those obtained in the present work and with those reported in the pertinent literature for the a-C:H films, sliding against various different counterparts [5]. The low friction coefficient and the wear response were attributed to the formation of the microcrystalline graphite-like wear debris that could stick to the mating surface. This fact was corroborated by Raman laser spectroscopy. These authors considered that this phenomena occurred as consequence of large flash temperatures at the contact points between the tribopairs and used the equation proposed by Ashby *et al.* [20] for evaluating this temperature, which resulted in a value of  $\Delta T \sim 300^\circ\text{C}$ .

On the other hand, since the flash temperature is directly proportional to the sliding velocity and in the present research this parameter is almost 6 times smaller, the value of  $\Delta T$  will be of approximately  $50^\circ\text{C}$ . As indicated by Bhushan [21], when studying the influence of sliding velocity and load on the wear of DLC coatings, this temperature rise is not sufficient to produce a phase transition process whereby a low shear strength graphite-like layer is formed at the sliding interface.

### 3.2 SEM morphology of the wear scars

The typical wear scar morphologies are shown in Figure 3 for the DLC/CrN/NiP/AA2024-T3 substrate as well as the alumina counterpart.

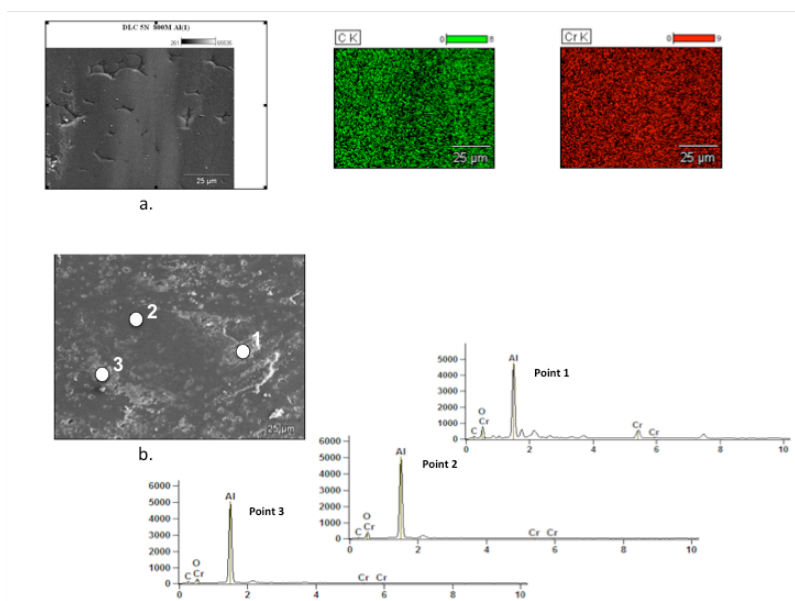


Figure 3. SEM micrograph of the wear scars: a. on the Al2024-T3 sample coated with DLC/CrC/NiP with the corresponding mapping of Cr and C; b. Alumina ball and the EDS analysis related to the points indicated in micrograph, showing the composition of the adhered material.  
(Normal load = 5 N, sliding velocity = 0.1 m/s and  $43 \times 10^6$  cycles)

As expected, the surface of the DLC coated system in the middle of the wear scar shows the typical nodular morphology, i.e., the cauliflower characteristic structure (Figure 3a) of the intermediate electroless layer of a higher thickness. The darker zone in the micrographs corresponds to the coating that has been worn out in a smaller amount, indicating that an uneven wear took place in contact with the alumina ball. This fact could be verified by the element mapping shown in the same Figure, where it is observed that in some regions of the wear scar the C distribution is not uniform.

Nevertheless, the Cr distribution is uniform all over the scar, which coupled with the fact that the friction coefficient value at the end of the test was of 0.1, indicates the preservation of the coating integrity, i.e. its existence, although as a very thin film, but without the presence of any surface failures.

Furthermore, the chemical analysis by EDS carried out to evaluate the morphology of the wear scar of the alumina counterpart used in this test indicated, as expected, the presence of some traces of carbon, chromium and oxygen at point 1 shown in this figure. These results demonstrate that the coating adhered to the aluminum ball as a consequence of the difference in the hardness values between the tribopairs. The presence of Cr is considered to be due to the existence of a small quantity of Cr in the coating, as a result of the manufacturing process, as it was mentioned previously.

### 3. CONCLUSIONS

The results indicate that, under the present testing conditions, the duplex coating of NiP/CrC-DLC deposited onto a AA2024-T3 substrate exhibits a satisfactory behavior from the mechanical stability point of view, when the thickness of the NiP layer is higher than 60  $\mu\text{m}$ , since no surface failures were observed at the end of the tests. In this way, the results of the analytical calculation of the elastic contact deformation fields under a spherical indenter, for the case of Hertzian contact employing the software Elastica©, have been verified. It was found that for the coated system, the magnitude of the friction coefficient was of approximately 0.1 and that

of the wear rate was of about  $2.31 \pm 0.09 \times 10^{-16} \text{ m}^3/\text{N.m}$ . On the contrary, for the uncoated substrate, the friction coefficient was of approximately 0.5 and the wear rate of  $5.46 \times 10^{-13} \text{ m}^3/\text{N.m}$ , that is to say, 3 orders of magnitude greater than that determined for the coated system.

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