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A volumetric energy based high cycle multiaxial fatigue criterion

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Abstract

A reliable design of industrial parts against high-cycle multiaxial fatigue requires a fatigue criterion capable of predicting both the stress gradient and the load-type effects. These effects are very important in the transfer of fatigue data from specimen to component. By using the concept of volume influencing fatigue crack initiation proposed by Palin-Luc and Lasserre with an energy based approach, a new criterion is presented. Based on the strain-work density given to the material, this proposal is usable whatever the constant amplitude loading is: in and out-of-phase combined loadings, with or without mean stress. Its predictions are compared both with a total of 38 experiments on four materials (a mild steel, two high strength steels and a spheroidal graphite cast iron) and with the predictions of local criteria (Crossland, Dang Van, Papadopoulos and Morel). The comparison shows that the predictions of the volumetric proposal are very good and less scattered than those of the local approaches, especially for loadings with mean stresses or under non-proportional loadings

Keywords: High cycle fatigue; Criterion; Multiaxial; Energy; Gradient; Volume

1. Introduction

Many high-cycle multiaxial fatigue criteria are proposed in the literature for metals. Four categories of fatigue criteria can be distinguished. First, until the end of the fifties several empirical formulae were proposed to synthesize many fatigue data (Haigh, Gerber, Marin, Gough and Pollard, etc...). Second, from the observation that after nucleation a microcrack propagates first along a shear plane, many authors assumed that crack initiation is governed by the shear stress (McDiarmid, Findley) or by the second invariant of the deviatoric stress tensor (Sines, Crossland). Third, after the criterion proposed by Dang Van [1], other micro-macro approaches appeared (Papadopoulos, Deperrois, Morel). They consider that elastic shakedown is the condition needed to avoid fatigue crack initiation in unfavorably oriented grains. Fatigue crack initiation in polycrystalline metal is determined from the critical plane containing the easiest

slip directions of the grain and experiencing the largest shear strain amplitude. Finally, some fatigue criteria are based on global energy quantities: elastic, plastic or total strain energy density. These approaches do not predict crack orientation, but the computation time of such a model is shorter than for critical plane criteria. All the above proposals are based on the stress (or strain) tensor at a critical point and none of them is able to predict the stress gradient and size effects, nor the load type effect (endurance limit in tension is not the same as in bending [2,3]). Furthermore the machining process also has an important effect on fatigue strength because fatigue crack nucleation depends on the finishing state of the component surface (grinding, polishing, etc...); in this paper, this “technological effect” is not considered. The surface layer and roughness of both specimen and component (with residual stresses if any) are presumed to be identical.

However, a reliable design of components against high-cycle multiaxial fatigue is still very difficult for engineers because the transfer of fatigue data from specimens to industrial parts cannot be done without considering the previous effects [4]. By analyzing fatigue data from Pogoretskii and Karpenko [5], and from Phillips

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Nomenclature

dT	triaxiality degree of stresses
\vec{n}	vector normal to a material plane
r	radius in cylindrical coordinates
t	time
A	fracture elongation in tension
C_i	fatigue critical point
E	Young modulus
$F(dT, \beta)$	correction function depending on the triaxiality degree of stresses
R	specimen radius
R_σ	load ratio $\sigma_{\min}/\sigma_{\max}$
REP	relative error of prediction
Rm	maximum tensile strength
$\underline{\underline{S}}(M, t)$	deviatoric part of the stress tensor at point M function of time
T	loading period
V^*	volume influencing fatigue crack initiation
W_g	strain work density given to the material on a loading period
W_g^*	threshold value of W_g corresponding to σ^*
β	material parameter of the volumetric energy based proposal
$\underline{\underline{\varepsilon}}(M, t)$	strain tensor at point M function of time
ν	Poisson ratio
ϖ_g	mean value over V^* of the damaging part of W_g
ϖ_g^D	value of ϖ_g at the endurance limit
$\varpi_{g, \text{Ten}}$	expression of ϖ_g in tension
$\varpi_{g, \text{RotBen}}$	expression of ϖ_g in rotating bending
$\varpi_{g, \text{PB}}$	expression of ϖ_g in plane bending
σ_a	normal stress amplitude
σ_{eq}	Von Mises equivalent stress
σ_H	hydrostatic stress
$\bar{\sigma}$	mean normal stress
σ^*	threshold stress below which there is no micro damage initiation
$\sigma_{\text{Ten}, -1}^D$	fully reversed endurance limit in tension
$\sigma_{\text{RotBen}, -1}^D$	fully reversed endurance limit in rotating bending
$\sigma_{\text{PB}, -1}^D$	fully reversed endurance limit in plane bending
σ_Y	yield stress in tension
τ_a	shear stress amplitude
$\bar{\tau}$	mean shear stress
τ_{-1}^D	fully reversed endurance limit in torsion
τ_Y	yield stress in torsion
ϕ	phase shift for combined loading
$\underline{\underline{\Sigma}}(M, t)$	stress tensor at point M function of time

and Heywood [6], Papadopoulos et al. show that the pure gradient effect, i.e. the volumetric stress distribution effect, is more important than the pure size effect [5,7]. Moreover, it is not enough to characterize a notch with the theoretical stress concentration factor K_t [8], and the fatigue notch factor K_f . For two components with different geometries these factors can be the same even if experimental fatigue limits are different [9]. All the previous fatigue criteria do not predict the pure gradient effect because they only consider stress and strain ten-

sors at the critical point. This is the reason why, from about twenty years ago, a few fatigue criteria aim to model the gradient effect: the critical layer of Flavenot and Skally [10], the Papadopoulos gradient dependent fatigue criterion [11], the effective distance of Pluvinage [9,12], the maximum stressed-strained volume of Sonsino [4], the volume influencing fatigue crack initiation proposed by Palin-Luc and Lasserre [3], and the Robert et al. gradient dependent criterion [13]. An interesting literature review of such models is presented in [14].

With a more general point of view, outside the field of fatigue research, many non-local failure criteria exist (Seweryn and Mroz [15], Kennedy and Nahan [16]).

This paper presents an evolution of the energy based volumetric fatigue criterion proposed by Palin-Luc and Lasserre [3]. The threshold stress defining the volume influencing fatigue crack initiation is discussed and the predictions are analyzed for proportional and non-proportional cyclic loadings, with or without mean stresses. The predictions of this proposal are presented and compared to experiments and to the predictions calculated with some local criteria. All the fatigue criteria predict the fatigue strength corresponding to fatigue crack initiation. In experiments the size of the nucleated fatigue crack is that which a fatigue testing machine can automatically detect by monitoring the specimen stiffness decreasing: about 0.5–1 mm in depth; crack propagation is not considered hereafter.

2. Proposed non-local criterion

In 1998, from experiments on a spheroidal graphite cast iron Palin-Luc and Lasserre [17] showed that a new limit σ^* can be defined below the usual endurance limit σ^D . At a considered point, a stress amplitude below this new limit does not initiate observable damage at the microscale (no micro-crack). Between σ^* and σ^D a stress amplitude only contributes to micro-damage initiation which could develop if, either near this point or in the course of time, there is a stress amplitude higher than the endurance limit. The usual endurance limit is a limit of no damage propagation but is not a limit of no damage initiation at the microscale. Several works proved that micro-cracks could initiate but not propagate to macro-cracks on a specimen loaded at its endurance limit. This stress limit σ^* can be estimated [3] from fatigue test results in fully reversed tension and in rotating bending by Eq. (1).

$$\sigma^* = \sqrt{2(\sigma_{\text{Ten}, -1}^D)^2 - (\sigma_{\text{RotBend}, -1}^D)^2} . \quad (1)$$

By using this threshold stress Palin-Luc and Lasserre [3] proposed a volumetric energy based high-cycle multiaxial fatigue criterion which is restricted to fully reversed loadings. It also has the main drawback of being load-shape dependent for uniaxial loading (sinus, triangle, square). This is not in agreement with the fatigue data of Mielke [18] and Dietmann et al. [19].

To avoid these lacks, we propose now to consider as damage parameter the strain work density given to each elementary volume of material per loading cycle, W_g (2). This assumption is based on the hypothesis (in agreement with thermodynamics [20]) that micro-damage cannot initiate and grow, from microscale to macroscale, without giving a strain work to each elementary volume of material [21].

$$W_g(M) = \sum_i \sum_j \int_T \langle \sigma_{ij}(M,t) \dot{\epsilon}_{ij}^e(M,t) \rangle dt \quad (2)$$

$$= \int_{\epsilon_{ij_{\min}}^e}^{\epsilon_{ij_{\max}}^e} \langle \sigma_{ij}(\epsilon_{ij}) d\epsilon_{ij}^e \rangle .$$

where $\langle a \rangle = a$ if $a > 0$ and $\langle a \rangle = 0$ if $a \leq 0$.

This parameter is equal to the integral over a loading cycle (of period T) of the positive part of the strain power density. It is also equal to the sum of the positive variations of the strain energy density over the loading period T (Fig. 1). W_g is calculated with the elastic strains after elastic shakedown of the material. At the endurance limit the stresses are low enough to consider that the material remains elastic at the macroscopic scale [20] after elastic shakedown. It has to be noticed that for a uniaxial stress state W_g is not dependent on the loading shape (sinus, triangle, square, etc ...) and is mean stress dependent. This will be discussed later.

In Fig. 1, between $t = 0$ and t_1 , a work is given to the material in tension; between t_1 and t_2 the material comes back to an equilibrium state (unloaded); between t_2 and t_3 , a work is given in compression; finally between t_3 and t_4 the material comes back to an unloaded state. Note that the strain-work density given by each stress and strain is summed independently, for instance in combined tension and torsion, the work given by σ_{xx} is summed with the work given by τ_{xy} (in Eq. (2) $\sum_i \sum_j \langle \sigma_{ij}(M,t) \dot{\epsilon}_{ij}^e(M,t) \rangle \neq \langle \sigma_{ij}(M,t) \dot{\epsilon}_{ij}^e(M,t) \rangle$). To illustrate this, let us consider a fully reversed combined tension (or bending) and torsion out-of-phase sinusoidal fatigue test with a phase shift of 90° , so that the ratio between the normal stress amplitude σ_a and the shear stress amplitude τ_a is: $\sigma_a/\tau_a = \sqrt{2(1+\nu)}$, where ν is the Poisson ratio. The non-null terms of the tensor of stresses are: $\sigma_{11}(t) = \sigma_a \sin(\omega t)$ and $\sigma_{12}(t) = \tau_a \sin(\omega t + \pi/2)$. For such a test, illustrated in Fig. 2A for $\sigma_a = 100 \sqrt{2(1+\nu)}$ MPa and $\tau_a = 100$ MPa, with a linear elastic isotropic material, the potential strain energy density is constant: $\sigma_{ij}(t) \epsilon_{ij}(t)/2 = (1+\nu)(\tau_a)^2/E$ as shown in Fig. 2B. This is due to the fact that during the time interval when a work is given to the material in tension, the strain energy due to torsion is released, and vice versa (Fig. 2B). But, if you compute during a loading period the strain power density due to tension, $\sigma_{11}(t) \dot{\epsilon}_{11}(t)$, and the corresponding part due to torsion, $2\sigma_{12}(t) \dot{\epsilon}_{12}(t)$, you can see in Fig. 2C that these quantities are not constant. Since we assume that a strain work has to be given to each elementary volume of material to create damage, the strain work density given to the material due to bending has to be summed with the strain work density given to the material due to torsion. This corresponds to the dashed area of the curve: strain power density versus time in Fig. 2C.

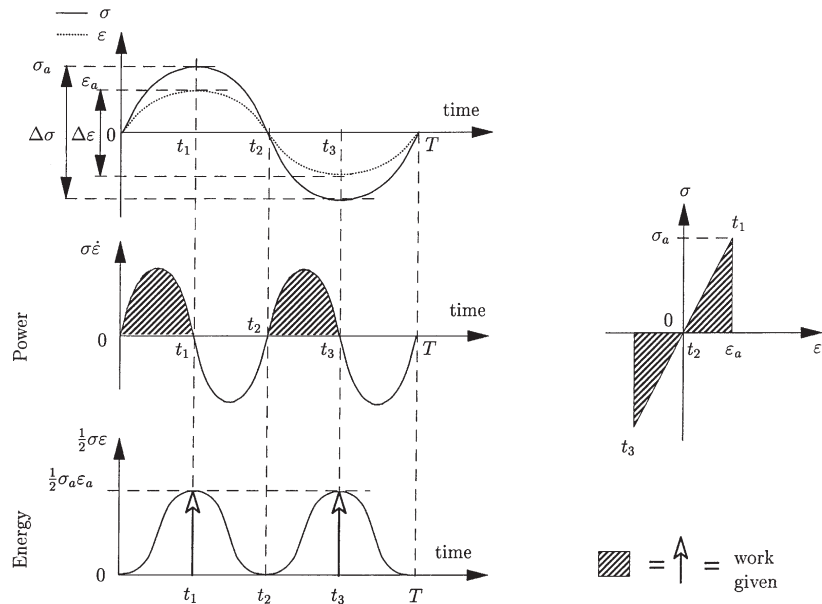


Fig. 1. Volumetric density of the elastic strain work and strain power given to the specimen on a loading period T in fully reversed tension.

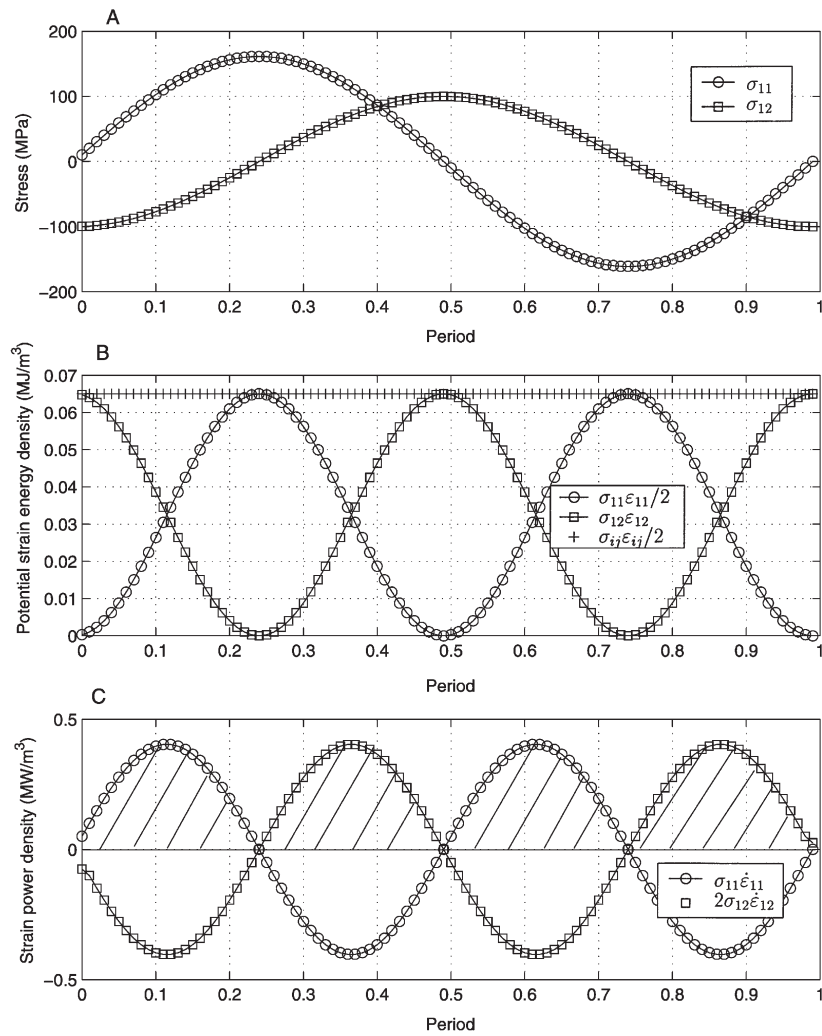


Fig. 2. Evolution versus time of the stresses (A), the potential strain energy density (B) and the strain power density (C) for a fully reversed combined tension and torsion out-of-phase sinusoidal fatigue test with a phase shift of 90° , so that the ratio between the normal stress amplitude σ_a and the shear stress amplitude τ_a is: $\sigma_a/\tau_a = \sqrt{2(1+\nu)}$.

2.1. Remark about the Findley's experiment

A well known paper by Findley [22] is often quoted in a way to debate energy based approaches in fatigue. This issue presents fatigue tests carried out on a rotating disk loaded in two perpendicular diameters by rollers. At the center of the disk, a cyclic biaxial stress state is imposed, where the non-null components of the tensor of stresses can be written as [23]: $\sigma_{11}(t) = 2.9\sigma \sin(\omega t) - 1.1\sigma$, $\sigma_{22}(t) = 2.9\sigma \sin(\omega t - \pi) - 1.1\sigma$, $\sigma_{12}(t) = 2.9\sigma \sin(\omega t - \pi/2)$ where σ is a constant proportional to the load. The amplitude of the principal stresses is constant but the principal stress directions are rotating compared with the axes $(\vec{x}_1, \vec{x}_2, \vec{x}_3)$ linked to the specimen. By calculating the time evolution of the potential strain energy density, $\frac{1}{2}\sigma_{ij}(t)\varepsilon_{ij}(t)$, at the center

of the disk where crack appeared, the author remarked that this variable is constant during a load period though the load is cyclic. Indeed, in this case for an isotropic material with a linear elastic behaviour $\frac{1}{2}\sigma_{ij}(t)\varepsilon_{ij}(t) = \frac{1}{E}[(1 + \nu)(2.9\sigma)^2 + (1 - \nu)(1.1\sigma)^2]$. Since a quasi-static tension test can generate the same strain energy density without fatigue crack initiation, Findley concluded that this variable cannot be used as a damage parameter in a fatigue criterion for combined loadings.

But it has to be pointed out that Findley computed the potential strain energy density and not the strain work density given per cycle to the material. As previously explained, this work is equal to the positive variation of the strain energy density over a loading period. For the stress state of the Findley's experiment, the strain power densities corresponding to each stress and strain term of the tensor of stresses and strains are not constant, as illustrated in Fig. 3 (example with $E = 200$ GPa, $\nu = 0.3$ and $\sigma = 100$ MPa). This shows that even if the potential strain energy density is constant some strain work is given to the material (it is impossible to create stresses and strains without giving a work to the material).

As proposed, by separating the strain works given by each term of the stress tensor, a non-null strain work density given to the material can be always calculated for a cyclic load. In a similar way, Park and Nelson [23] show that the distortion work is not null for the Findley experiments.

2.2. Volume influencing fatigue crack initiation

As explained before, fatigue tests in blocks carried out on smooth specimens in spheroidal graphite (SG) cast iron, in fully reversed plane bending and torsion [24,25], prove that blocks with a stress amplitude below the conventional endurance limit and above a threshold stress

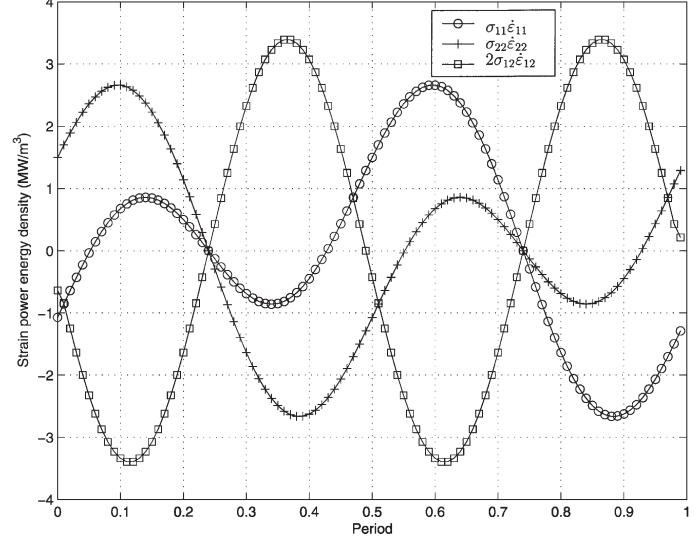


Fig. 3. Evolution of the different power strain energy densities $\sigma_{11}\dot{\varepsilon}_{11}$, $\sigma_{22}\dot{\varepsilon}_{22}$, $2\sigma_{12}\dot{\varepsilon}_{12}$, corresponding to each non-null products $\sigma_{ij}\dot{\varepsilon}_{ij}$ versus time during a loading period of the Findley's experiments.

amplitude, noted σ^* , participate in damage. SEM observations of a specimen loaded at a stress level higher than this limit, show quick micro-crack initiation which does not propagate if the stress amplitude stays below the usual endurance limit. More recently, sinusoidal plane bending fatigue tests were carried out on specimens with a square cross-section made in the same material. SEM observations of the specimen's lateral faces, where there is a stress gradient, show micro-damaged areas. Micro-cracks were observed in areas loaded at stress levels above a threshold stress which corresponds well to the threshold identified from experiments by blocks σ^* [21]. Then, these observations show that there is a micro-damaged volume around the fatigue critical point. With the hypothesis that all this damaged volume, noted V^* , participates in macro-crack initiation (engineer scale), a criterion has to take into account stress and strain distribution inside this volume.

As proposed by Palin-Luc and Lasserre [3] we consider a volume influencing fatigue crack initiation V^* to predict the effects, on the fatigue strength, of the stress-strain distribution inside the component. For a fully reversed uniaxial stress state, this volume is defined by the set of points loaded by a stress amplitude higher than a material dependent threshold stress σ^* . For fully reversed sinusoidal load,¹ W_g is proportional to the mean value of the strain energy density W_a [21]. The threshold work W_g^* corresponding to the stress limit σ^* is defined. The stress-strain distribution effect is taken into account by computing the mean value, $\bar{\omega}_g$, over the volume V^* (4), of the strain work given to the material exceeding the threshold W_g^* . This is the damaging part of the strain work density given per cycle to the speci-

¹ Except for out of phase biaxial push-pull loads.

men (3) around the critical point C_i (where W_g has a local maximum in the structure). The fatigue criterion is then expressed by Eq. (5) where $\varpi_g^D(C_i)$ is the value of $\varpi_g(C_i)$ at the endurance limit.

$$\varpi_g(C_i) = \frac{1}{V^*(C_i)} \iiint_{V^*(C_i)} [W_g(M) - W_g^*] dv. \quad (3)$$

$$V^*(C_i) = \{ \text{points } M(x,y,z) \text{ around } C_i \text{ so that } W_g(M) \geq W_g^* \}. \quad (4)$$

$$\varpi_g(C_i) \leq \varpi_g^D(C_i). \quad (5)$$

2.3. Uniaxial stress states

For uniaxial stress states, the hypothesis that at the endurance limit $\varpi_{g,\text{uniax}}^D$ is constant whatever the loading (tension, rotating bending, plane bending), $\varpi_{g,\text{uniax}}^D = \varpi_{g,\text{Ten}}^D = \varpi_{g,\text{RotBen}}^D = \varpi_{g,\text{PB}}^D$, gives the relations needed to identify the thresholds σ^* and W_g^* from both the fully reversed endurance limits in tension and rotating bending on smooth specimens (8). These two loadings are chosen as reference because of the same W_g distributions on the cross-section except along the specimen radius. As there is no stress gradient along the longitudinal axis of a cylindrical smooth specimen, the volume V^* can be reduced to a surface S^* inside the cross section of the specimen (Fig. 4). Eq. (3) applied to the fully reversed tension loading becomes Eq. (6). S^* is all the cross-section, since all the points are loaded at the same stress level.

$$\varpi_{g,\text{Ten}}^D = (\sigma_{\text{Ten},-1}^D)^2 / E - W_g^*. \quad (6)$$

In four point rotating bending, S^* is a crown between

the radius r^* of the circle representing the iso- W_g^* line, and the specimen radius R . Eq. (3) applied to this case becomes Eq. (7) where $(r^*/R) = \sigma^* / \sigma_{\text{RotBen},-1}^D$

$$\varpi_{g,\text{RotBen}}^D(C_i) = \frac{(\sigma_{\text{RotBen},-1}^D)^2}{2E} [1 - (r^*/R)^2]. \quad (7)$$

From Eqs (6) and (7) the material parameters W_g^* and σ^* are identified (8) by noting that $W_g^* = (\sigma^*)^2 / E$.

$$W_g^* = \frac{2(\sigma_{\text{Ten},-1}^D)^2 - (\sigma_{\text{RotBen},-1}^D)^2}{E} \text{ and} \quad (8)$$

$$\sigma^* = \sqrt{2(\sigma_{\text{Ten},-1}^D)^2 - (\sigma_{\text{RotBen},-1}^D)^2}.$$

where $\sigma_{\text{Ten},-1}^D$, and $\sigma_{\text{RotBen},-1}^D$ are the endurance limits in fully reversed tension and rotating bending on a smooth specimen.

At the present stage of the model development it is assumed that the threshold work W_g^* does not depend on the mean stresses. Then, the value of $\varpi_{g,\text{uniax}}^D$ is expressed as follows.

$$\varpi_{g,\text{uniax}}^D = \frac{(\sigma_{\text{RotBen},-1}^D)^2 - (\sigma_{\text{Ten},-1}^D)^2}{E}. \quad (9)$$

The endurance limit can then be predicted for a cylindrical specimen loaded in four-point plane bending. Only two points are critical on the specimen cross-section where the distribution of the elastic strain work given

$$\text{per cycle is: } W_{g,\text{PIBen}}(M) = \frac{(\sigma_{\text{PIBen},-1}(C_i))^2}{E} \left(\frac{y}{R} \right)^2.$$

The iso- W_g^* lines are defined by the distance y^* from the neutral fiber (O, \vec{x}). The surface $S^*(C_i)$ around the two critical points C_i is illustrated by Fig. 5 and defined by Eq. (10), where $\alpha = \arcsin \left(\frac{y^*}{R} \right)$.

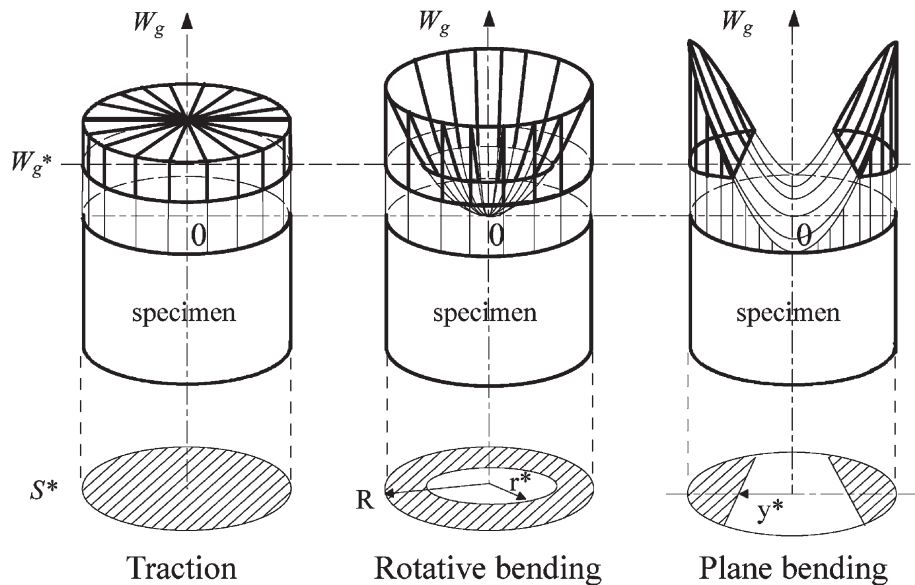


Fig. 4. Distribution of the volumetric density of the strain work given per cycle on the cross section of a smooth cylindrical specimen in tension, rotating bending and plane bending.

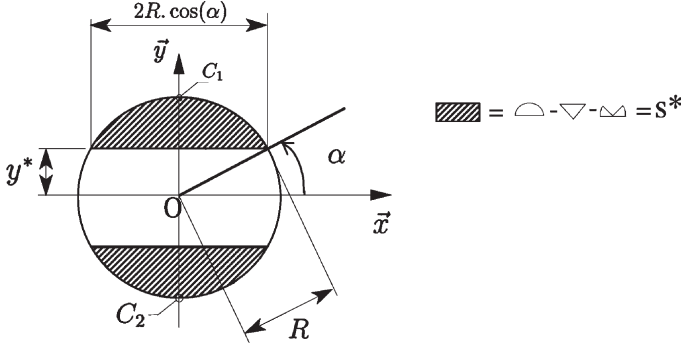


Fig. 5. Surfaces $S^*(C_1)$ and $S^*(C_2)$ on the cross-section of a smooth specimen loaded in four point plane bending.

$$\left(\frac{y^*}{R}\right)^2 = \frac{\sigma_{\text{PIBen},-1}^2(C_i)}{\sigma^{*2}} \text{ and} \quad (10)$$

$$S^*(C_i) = \left(\pi - \frac{2y^* \cos(\alpha)}{R} - 2\alpha\right) \frac{R^2}{2}.$$

Then, the endurance limit σ_{PIBen}^D is the solution of equation $\varpi_{g,\text{PIBen}}^D(C_i) = \varpi_{g,\text{uniax}}^D$ with,

$$\varpi_{g,\text{PIBen}}^D(C_i) = \frac{(\sigma_{\text{PIBen},-1}^D(C_i))^2}{E} [A(\theta) - \theta^2]. \quad (11)$$

where θ and $A(\theta)$ are given by Eq. (12).

$$\theta = \frac{y^*}{R} \text{ and} \quad (12)$$

$$A(\theta) = \frac{\frac{\pi}{8} - \frac{\theta}{4}(2\theta^2 - 1)\sqrt{1 - \theta^2} - \frac{\arcsin\theta}{4}}{\frac{\pi}{2} - \theta\sqrt{1 - \theta^2} - \arcsin\theta}.$$

2.4. Multiaxial stress states

Fatigue crack initiation depends on the triaxiality of stresses. Indeed, the endurance limit in tension is not the same as in torsion, for instance. Furthermore, depending on its ductility, each material does not have the same sensitivity to the triaxiality of stresses. Thus, the material sensitivity to the triaxiality of stresses is taken into account in our proposal by using the empirical function F , proposed by Froustey et al. [26] and already used by Palin-Luc and Lasserre [17]. This function is defined by Eq. (14) where $dT(M)$ is the triaxiality degree defined by the ratio of the spherical part of the strain work given to the material W_g^s over W_g (13). dT varies between 0 (pure torsion) and 1 (equi-axial tension or compression). β is a material parameter. As proposed by Palin-Luc [3], this function is the ratio, at the endurance limit, of the strain work volumetric density given to the material whatever the load is, W_g^D , over the same value in fully reversed torsion $W_{g,\text{Tors},-1}^D$

$$dT(M) = \frac{W_g^s(M)}{W_g(M)} \text{ and } W_g^s(M) = \int_T \langle \sigma_{kk} \dot{\epsilon}_{kk} \rangle dt. \quad (13)$$

$$F(dT(C_i), \beta) = \frac{W_g^D(C_i)}{W_{g,\text{Tors},-1}^D} = \frac{1}{1 - dT(C_i)} \left\{ 1 - \frac{1}{\beta} \ln [1 + (e^\beta - 1) dT(C_i)] \right\}. \quad (14)$$

The material parameter β can be identified from two endurance limits under loads generating two different stress states. The fully reversed endurance limits on smooth cylindrical specimen in four point rotating bending $\sigma_{\text{RotBen},-1}^D$ and in torsion τ_{-1}^D are recommended, because for both these loads there is only a stress gradient along the radius of the specimen. From Eq. (14) for these two loadings, β is solution of Eq. (15) where ν is the Poisson ratio of the material.

$$\left(\frac{\sigma_{\text{RotBen},-1}^D}{\tau_{-1}^D}\right)^2 - 3 \left\{ 1 - \frac{1}{\beta} \ln \left[1 + \frac{1-2\nu}{3} (e^\beta - 1) \right] \right\} = 0. \quad (15)$$

For a very ductile material β is around -1 (C20 annealed steel for instance); for a brittle one (EN-GJS800-2 SG cast iron) β is around 3. It has to be noticed that for a ductile material $F \approx 1$ whatever the degree of triaxiality, so the relation between the endurance limits which are predicted by the model in torsion and in rotating bending is close to the CETIM empirical proposal for steels [27]: $\tau_{-1}^D \approx 0.6 \times \sigma_{\text{RotBen},-1}^D$.

By hypothesis and as used by [3] this function $F(dT(C_i), \beta)$ is also used to compute the threshold strain work W_g^* for different triaxiality degrees. From Eq. (14) it is easy to show Eq. (16), where $dT_{\text{uniax}} = (1 - 2\nu)/3$ is the triaxiality degree for a uniaxial stress state. The volume influencing fatigue crack initiation V^* is thus defined whatever the stress state is at the critical point (17).

$$W_g^*(C_i) = \frac{F(dT(C_i), \beta)}{F(dT_{\text{uniax}}, \beta)} W_{g,\text{uniax}}^*. \quad (16)$$

$$V^*(C_i) = \{ \text{points } M(x,y,z) \text{ around } C_i \text{ so that } W_g(M) \geq W_g^*(C_i) \}. \quad (17)$$

Finally, it is assumed that the value of $\varpi_g(C_i)$ at the endurance limit, noted $\varpi_g^D(C_i)$, depends on the triaxiality degree of stresses according to Eq. (18).

$$\varpi_g^D(C_i) = \frac{F(dT(C_i), \beta)}{F(dT_{\text{uniax}}, \beta)} \varpi_{g,\text{uniax}}^D. \quad (18)$$

The criterion can then be applied whatever the stress state is, by comparing the value of $\varpi_g(C_i)$, see Eq. (19), and its previous limit value for the corresponding triaxiality degree of stresses at C_i (18). Fatigue crack initiates before long life if $\varpi_g(C_i) \geq \varpi_g^D(C_i)$.

$$\varpi_g(C_i) = \frac{1}{V^*(C_i)} \iiint_{v^*(C_i)} [W_g(M) - W_g^*(C_i)] dv. \quad (19)$$

2.5. Case study example: combined tension and torsion on smooth specimen

To illustrate how to apply this criterion, let us study a combined tension and torsion fatigue test on a cylindrical smooth specimen with mean loads in tension $\bar{\sigma}$ and in torsion $\bar{\tau}$ so that $\sigma_a > \bar{\sigma}$ and $\tau_a > \bar{\tau}$. At the point $M(r, \theta, \phi)$ on a specimen cross-section, the tensor of stresses is given by Eq. (20) in cylindrical coordinates, where ϕ is the phase shift between the shear stress due to torsion and the normal stress due to tension. The distribution of the strain work density given to the material is written in Eq. (21).

$$\underline{\underline{\Sigma}}(M, t) = \begin{pmatrix} 0 & 0 & 0 \\ 0 & 0 & [\bar{\tau} + \tau_a \sin(\omega t + \phi)] \frac{r}{R} \\ 0 & [\bar{\tau} + \tau_a \sin(\omega t + \phi)] \frac{r}{R} & [\bar{\sigma} + \sigma_a \sin(\omega t)] \end{pmatrix}_{(e_r, e_\theta, e_z)} \quad (20)$$

$$W_{g, \text{Ten}+\text{To}}(M) = \frac{\bar{\sigma}^2 + \sigma_a^2}{E} + \frac{\bar{\tau}^2 + \tau_a^2}{E} 2(1 + \nu) \left(\frac{r}{R}\right)^2.$$

In this equation, R is the radius of the specimen and r the radius at point M . Critical points C_i are located at the circumference of the specimen ($r = R$). The spherical part of the given strain work density W_g^s at a critical point is:

$$W_{g, \text{Ten}+\text{To}}^s(C_i) = \frac{1-2\nu}{3E} (\bar{\sigma}^2 + \sigma_a^2). \quad (22)$$

The triaxiality degree of stresses $dT_{\text{Ten}+\text{To}}$ at each critical point is given by Eq. (23).

$$dT_{\text{Ten}+\text{To}}(C_i) = \frac{W_{g, \text{Ten}+\text{To}}^s(C_i)}{W_{g, \text{Ten}+\text{To}}(C_i)} \quad (23)$$

$$= \frac{1-2\nu}{3} \left\{ \frac{\bar{\sigma}^2 + \sigma_a^2}{\bar{\sigma}^2 + \sigma_a^2 + 2(1 + \nu)(\bar{\tau}^2 + \tau_a^2)} \right\}.$$

The energy based threshold value $W_{g, \text{Ten}+\text{To}}^*$ is calculated from Eq. (16) and is expressed in Eq. (24).

$$W_{g, \text{Ten}+\text{To}}^*(C_i) \quad (24)$$

$$= \left[\frac{2 \cdot (\sigma_{\text{Ten}, -1}^D)^2 - (\sigma_{\text{RotBen}, -1}^D)^2}{E} \right] \frac{F(dT_{\text{Ten}+\text{To}}(C_i), \beta)}{F(dT_{\text{uniax}}, \beta)}.$$

Then the volumetric mean value of the damaging strain work given to the material in the volume influencing fatigue crack initiation around the critical point $\varpi_{g, \text{Ten}+\text{To}}(C_i)$ is computed from Eq. (19) with Eqs. (21) and (24). The following equation is obtained:

$$\varpi_{g, \text{Ten}+\text{To}}(C_i) = \frac{2(1 + \nu)(\bar{\tau}^2 + \tau_a^2)}{2E} \left(1 + \left(\frac{r^*}{R}\right)^2 \right) \quad (25)$$

$$+ \frac{\bar{\sigma}^2 + \sigma_a^2}{E} W_{g, \text{Ten}+\text{To}}^*(C_i)$$

where the radius r^* limiting $V^*(C_i)$ is solution of Eq. (26) obtained from $W_{g, \text{Ten}+\text{To}}(C_i) = W_{g, \text{Ten}+\text{To}}^*$. Note that r^* must verify $0 \leq r^* \leq R$.

$$\left(\frac{r^*}{R}\right)^2 = \left[\frac{2 \cdot (\sigma_{\text{Ten}, -1}^D)^2 - (\sigma_{\text{RotBen}, -1}^D)^2}{2(1 + \nu)(\bar{\tau}^2 + \tau_a^2)} \right] \frac{F(dT_{\text{Ten}+\text{To}}(C_i), \beta)}{F(dT_{\text{uniax}}, \beta)} \quad (26)$$

$$= \frac{\bar{\sigma}^2 + \sigma_a^2}{2(1 + \nu)(\bar{\tau}^2 + \tau_a^2)}.$$

Then, to avoid fatigue crack initiation at the critical point, the load variables $\bar{\sigma}$, σ_a , τ and τ_a have to verify the relations (27) and (28):

$$\text{If } 0 < \left(\frac{r^*}{R}\right) < 1 \Rightarrow [2(1 + \nu)(\bar{\tau}^2 + \tau_a^2) + \bar{\sigma}^2] + \sigma_a^2 \left[\frac{F(dT_{\text{uniax}}, \beta)}{F(dT_{\text{Ten}+\text{To}}(C_i), \beta)} \right] < (\sigma_{\text{RotBen}, -1}^D)^2. \quad (27)$$

$$\text{If } r^* = 0 \Rightarrow [(1 + \nu)(\bar{\tau}^2 + \tau_a^2) + \bar{\sigma}^2] + \sigma_a^2 \left[\frac{F(dT_{\text{uniax}}, \beta)}{F(dT_{\text{Ten}+\text{To}}(C_i), \beta)} \right] < (\sigma_{\text{Ten}, -1}^D)^2. \quad (28)$$

The predictions of the proposed criterion, according to Eqs (27) and (28), are illustrated in Fig. 6 for different mean stresses on a non-dimensional diagram. These predictions do not depend on the phase shift ϕ between ten-

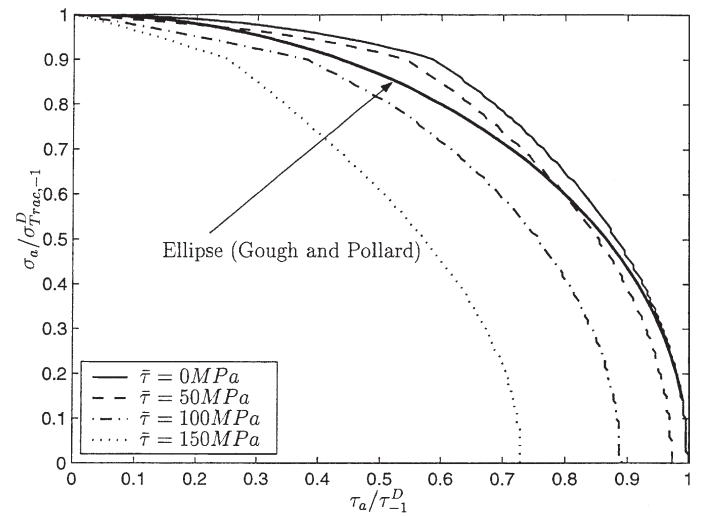


Fig. 6. Prediction of the proposal in combined tension and torsion for different mean shear stress (example for the EN-GJS800-2 SG cast iron, $\beta = 3.09$).

sion and torsion, which is in agreement with experimental results of the tests carried out by Simburger et al. [28] and Heidenrich et al. [29] on a middle ductile steel. For a combined loading without mean shear stress, the predictions are close to an ellipse arc similar to the Gough and Pollard proposal in combined bending and torsion. If there is no alternating shear stress, the mean shear stress influence on the criterion predictions is very low. This is in agreement with Froustey's experiments on the 30NiCrMo16 steel [30].

2.6. Synoptic of the proposal

The use of this criterion on an industrial component requires a finite element analysis to compute W_g at all the points of the component from the stress and strain time history and to compute the volume influencing fatigue crack initiation. For a given long time, its application can be organized as illustrated on the flow chart in Fig. 7.

3. Discussion

3.1. Mean load effect

The strain work volumetric density given per cycle to the material W_g depends on the mean load (Fig. 8). For example, in uniaxial tension, $W_g(M)$ is equal to:

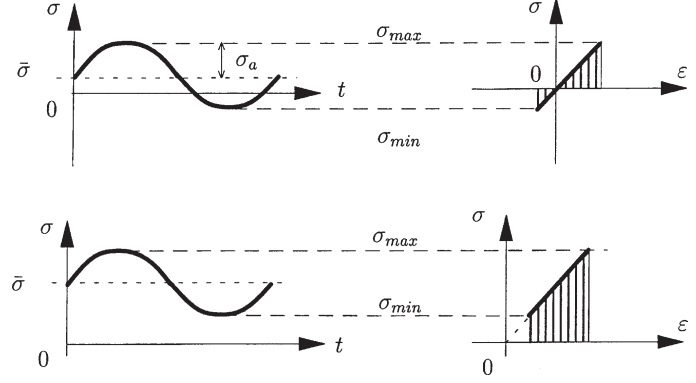


Fig. 8. Volumetric density of the elastic strain work given per cycle to the material for different mean loads in tension.

$$W_g(M) \quad (29)$$

$$= \begin{cases} \frac{\bar{\sigma}^2 + \sigma_a^2}{E} = \frac{\sigma_{\min}^2 + \sigma_{\max}^2}{2E} & \text{if } \sigma_a \geq |\bar{\sigma}| \Leftrightarrow R_\sigma \leq 0 \\ \frac{2\bar{\sigma}\sigma_a}{E} = \frac{|\sigma_{\min}^2 - \sigma_{\max}^2|}{2E} & \text{if } \sigma_a \leq |\bar{\sigma}| \Leftrightarrow R_\sigma \geq 0 \end{cases}$$

In a Haigh diagram, the criterion predictions depends on the material. Fig. 9 shows these predictions for the SG cast iron (EN-GJS800-2), a middle ductile material (quenched and tempered 30NiCrMo16 steel) and a ductile material (annealed C20 steel). In this diagram the curve illustrating the criterion has an inflexion point. At this point the tangent line is defined by $\sigma_{\max} = \text{constant}$. This constant corresponds approximately to the material yield stress σ_Y as shown in Fig. 9 for the 30NiCrMo16 steel. Thus, if the material is loaded below its yield stress, the effect of a normal mean stress on the predictions in push-pull tests is between the Goodman and the Gerber empirical criteria. This is in a good

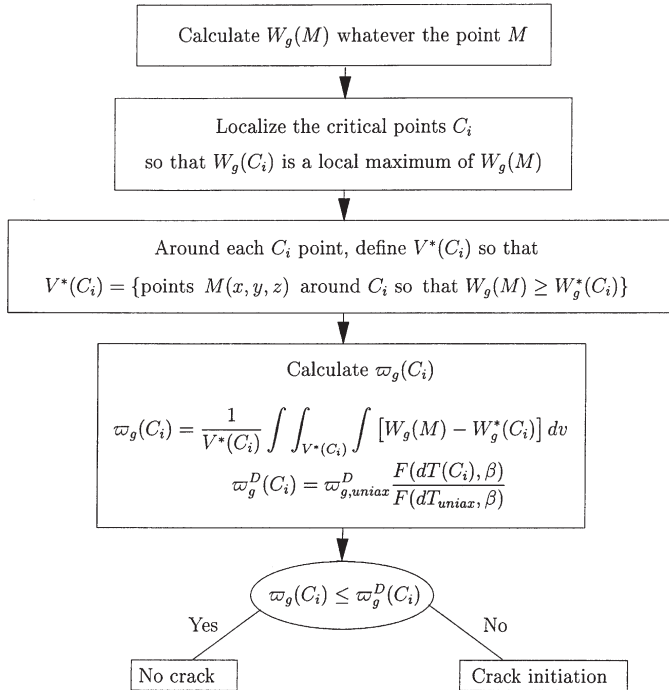


Fig. 7. Flow chart to apply the proposed non-local multiaxial fatigue criterion.

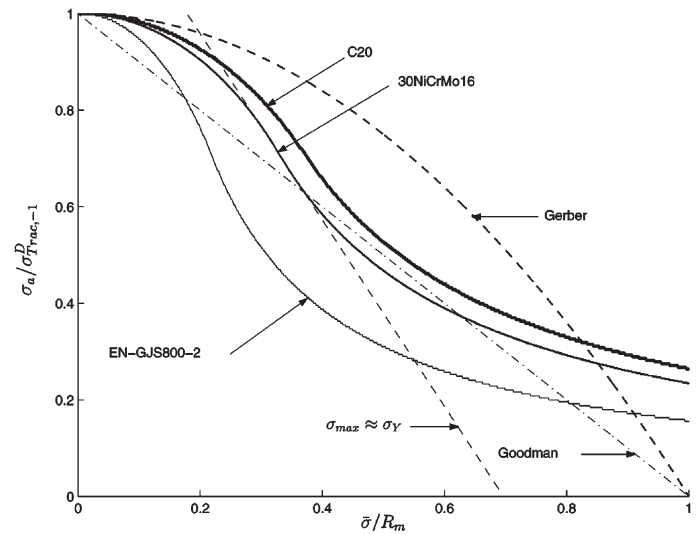


Fig. 9. Effect of mean normal stress on the criterion predictions in push-pull loadings in a Haigh diagram for three materials (example of the $\sigma_{\max} \approx \sigma_Y$ line for 30NiCrMo16 steel).

agreement with many fatigue data on different materials as reported by Smith [31], Sines [32] and Brand et al. [27]. These latter authors limit the Haigh diagram by a straight line defined by $\sigma_{\max} = \sigma_Y$.

Fig. 10 illustrates the endurance limits predicted by the proposal in pure torsion for the C20 steel, the 30NiCrMo16 steel and the SG cast iron EN-GJS800-2. For these three materials the predictions are in very good agreement with the observations of Sines [32], who pointed out the low influence of the mean torsion load if the maximum shear stress is below the shear yield stress τ_Y . It is also interesting to note that the predicted limit also depends on the material: the more brittle the material, the greater is the importance of its mean shear stress sensitivity. This point is confirmed by Smith [31].

In combined plane bending and torsion the proposed criterion is close to the ellipse arc of Gough and Pollard [33] as shown in Fig. 11. It has to be pointed out that there is a very small effect, on the predicted endurance limit, of a mean shear stress without alternating shear stress. This is in agreement with Froustey's experiments in high cycle fatigue [34].

3.2. Effect of the phase shift in combined loadings

There is a *phase shift effect* on the criterion predictions in *biaxial tension* as shown in Fig. 12. This is in agreement with experimental results carried out by Dietman et al. [19] on a St35 steel (EN-S185). These authors showed that, for ductile materials, there is a negative effect of the phase shift on the endurance limit, whereas for brittle materials this influence is positive: endurance limit for out of phase loadings is greater than for loading in phase. These experimental facts are well predicted by

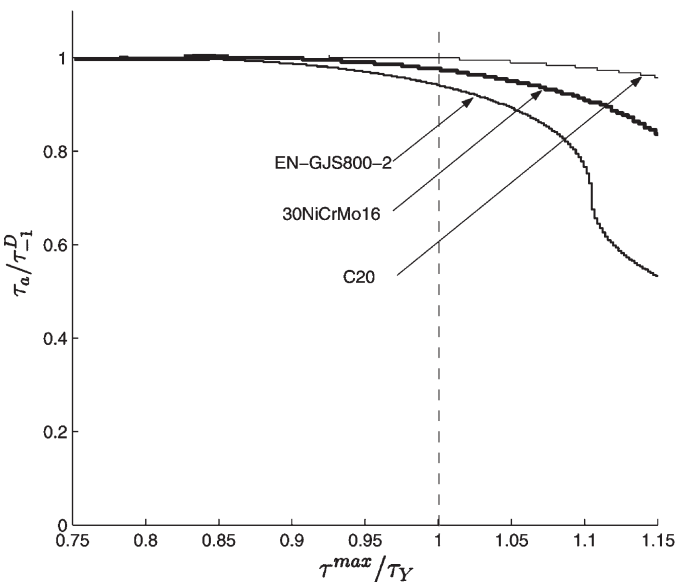


Fig. 10. Maximum shear stress influence on the allowable shear stress amplitude for the proposed non-local criterion (smooth specimen in torsion).

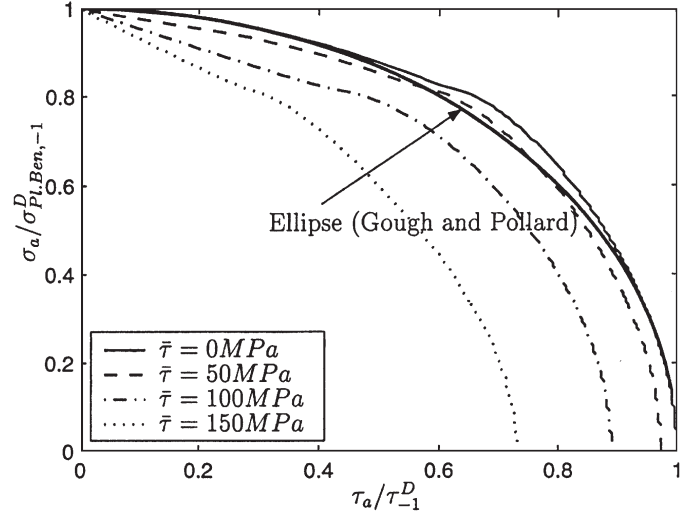


Fig. 11. Criterion predictions in combined plane bending and torsion on smooth specimen, for different mean shear stress (example for the EN-GJS800-2 SG cast iron).

our proposal, as illustrated in Fig. 12, with two types of materials: ductile and brittle. The different phase shift effects are linked with the value of the β parameter.

On the other hand, as explained in the case study presented before, *the proposal is not phase shift dependent in combined tension and torsion or bending and torsion*. This point is discussed in the literature by Sonino and Kueppers [35]. According to these authors:

- for ductile materials (structural steels, welded steels) out-of-phase loadings cause a reduction of the fatigue strength;
- for semi-ductile materials (cast steels, welded aluminum joints), the fatigue behaviour is neutral; and
- for brittle materials (cast iron, cast aluminum, sintered steels), out-of-phase loading increases the fatigue strength.

Nevertheless, this phase shift effect is small at the endurance limit (around 7% of the endurance limit for the EN-GJS800-2 SG cast iron [3]). The proposal predictions are in agreement with fatigue data in endurance if the scatter of experiments is considered [28,34].

3.3. Load type effect

The mean value, over the volume influencing fatigue crack initiation, of the volumetric strain work density given to the material, $\bar{\omega}_g$, is load type dependent; it also depends on the specimen geometry. So this parameter is different for all the uniaxial loadings: tension, rotating bending, four-point plane bending, three-point plane bending. The criterion takes into account the stress and strain volumetric distribution around the critical point, that is why it is sensitive to the load type. As shown in Fig. 13, with an example of EN-GJS800-2 SG cast iron,

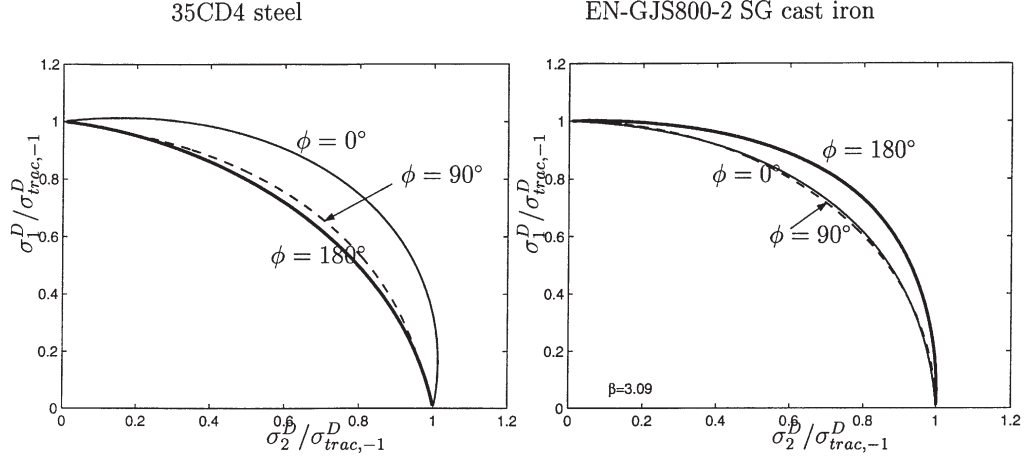


Fig. 12. Influence of the phase shift on the criterion prediction in biaxial tension for two materials: 35CD4 steel ($\beta = 1.33$) and EN-GJS800-2 SG cast iron ($\beta = 3.09$).

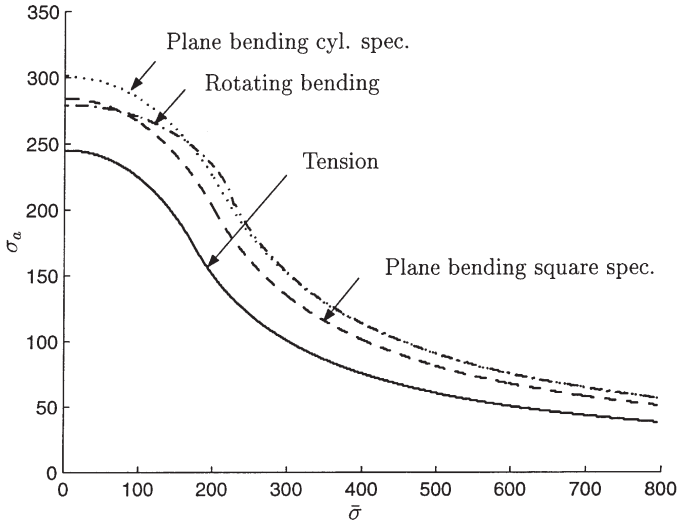


Fig. 13. Criterion predictions in a Haigh diagram in tension, plane bending on specimen with square cross section, with round cross-section and in rotating bending (example on EN-GJS800-2 cast iron).

the predictions distinguish the load type and the specimen geometry. This is in agreement with experiments [3]: the fully reversed endurance limit in tension on a smooth specimen is lower than in rotating bending; this last limit is also lower than in plane bending. Furthermore, the endurance limit in plane bending on a smooth specimen with a round cross-section is not the same as that on a smooth specimen with a square cross-section. Due to the fact that there is only a stress gradient along the radius of a round specimen loaded in rotating bending, the endurance limit in rotating bending is near the plane bending endurance limit on a specimen with square cross-section [24]. This is well predicted by our proposal as shown on the vertical axis in Fig. 13. Under combined loadings, depending on both the triaxiality degree of stresses and the material parameter β , the endurance limits predicted are also different.

4. Comparison between predictions and experiments

The accuracy of our proposal is compared with the accuracy of other well known local multiaxial fatigue criteria: Crossland [36], Dang-Van [37] (last proposal with double maximum on the plane orientation and in time), Papadopoulos [38] and Morel [39,40]. A short description of these models is presented in the Appendix.

The predictions of our proposal are compared with high-cycle multiaxial fatigue data (10^6 cycles or more) carried out on smooth specimens made of three materials: a quenched and tempered 30NiCrMo16 steel (from Froustey et al. [30]), an annealed C20 steel (from Galtier [41]) and a SG cast iron, EN-GJS800-2 (from Bennebach [42] and Palin-Luc [24]). Their main mechanical characteristics are given in Table 1; the endurance limits (at 10^6 cycles or more) of this table were obtained by the stair-case method with at least 15 specimens.

The identification of the parameters of each criterion was done by using the endurance limits in Table 1. These endurance limits were used to identify the parameters of our proposal σ^* and β . The value of the ratio $\sigma^* / \sigma_{Ten,-1}^D$ is material dependent. For the tested materials this ratio is between 0.70 and 0.90: for the 30NiCrMo16 steel $\sigma^* = 441$ MPa and $\beta = 0.96$, for C20 steel $\sigma^* = 230$ MPa and $\beta \approx -1.45$, for the EN-GJS800-2 SG cast iron $\sigma^* = 204$ MPa and $\beta = 3.09$. For the local criteria, the endurance limit in fully reversed torsion was always used as a reference. The reference uniaxial endurance limit was $\sigma_{Ten,-1}^D$ for the tests No. 1, 2 and 3 because these are experiments in tension; whereas for all the other tests the endurance limit in fully reversed plane bending $\sigma_p^{Ben,-1}$ was used. The M_σ approach was used for the Papadopoulos criterion. For the Morel proposal, a $S-N$ curve is needed to identify the p , q , and r parameters. The following values were identified by Banville [21]: (i) for EN-GJS800-2: $p = 70,000$ cycles, $q =$

Table 1
Mechanical characteristics (static and in high cycle fatigue) of the tested materials

Material	E (GPa)	ν	$\sigma_{y0.2}$ (MPa)	R_m (MPa)	A (%)	τ_{-1}^D (MPa)	$\sigma_{Ten,-1}^D$ (MPa)	$\sigma_{RotBen,-1}^D$ (MPa)	$\sigma_{Pl.Ben,-1}^D$ (MPa)
30NiCrMo16	200	0.29	950	1200	-	428	560	658	690
C20	210	0.3	350	520	24	186	273	310	332
EN-GJS800-2	165	0.275	462	795	9	220	245	280	294

8000 cycles and $r = 0$, (ii) for C20: $p = 600$ cycles, $q = 6000$ cycles and $r = 0$, (iii) for 30NiCrMo16: $p = 40,000$ cycles, $q = 4000$ cycles and $r = 0$.

A total of 38 comparisons between experiments and predictions are given in Table 2 by using the Relative Error of Pre-

diction REP(%) defined by Eq. (30). This was computed for a given mean load and a fixed loading ratio σ_a^D/σ_a^D . If REP is positive, the prediction is conservative (safety), i.e. prediction is lower than experiment. If REP is negative, the prediction is unconservative (it is greater than experimental result).

Table 2
Experimental fatigue data (in MPa) and REP values (%) for the tested fatigue criteria

No.	Material	Load	σ_a^D	$\bar{\sigma}^D$	τ_a^D	$\bar{\tau}^D$	$\phi(^{\circ})$	REP ^a Prop.	DV	Pap.	Cros.	Mo.
1	30NiCrMo16	Ten	235	745	0	0	0	10.2	-28.5^b	-60.8	-60.8	-29.3
2	30NiCrMo16	Ten	251	704	0	0	0	11.1	-26.3	-54.6	-51	-26.7
3	30NiCrMo16	Ten	527	222	0	0	0	2.3	8.3	4	27.9	7.8
4	30NiCrMo16	P1.B	575	375	0	0	0	-4	-8.8	-18.3	-18.3	-9.3
5	30NiCrMo16	P1.B	575	375	0	0	0	-7.3	-7.3	-15.7	-15.5	-8
6	30NiCrMo16	P1.B	627	273	0	0	0	-6.5	-1.6	-7	-7	-2.1
7	30NiCrMo16	P1.B	679	156	0	0	0	-3.8	3	0	0	2.2
8	EN-GJS800-2	P1.B	184	225	0	0	0	-10.3	-19	-25	-32	-12.5
9	30NiCrMo16	P1.B.+To	519	0	291	0	0	-8.1	5.4	2.7	2.7	4.8
10	30NiCrMo16	P1.B.+To	514	0	288	0	90	-9.1	-36.4	1.7	-34.2	-7.6
11	30NiCrMo16	P1.B.+To	451	294	250	191	0	-3.3	-0.4	-9.3	-9.3	-1.1
12	30NiCrMo16	P1.B.+To	462	294	258	191	90	-0.4	-39.6	-6.3	-45	-10.2
13	30NiCrMo16	P1.B.+To	474	294	265	0	45	-8.4	-3.4	-3.6	-11.6	-2.3
14	30NiCrMo16	P1.B.+To	464	294	259	0	60	-10.8	-13.15	-5.8	-21.1	-8.2
15	30NiCrMo16	P1.B.+To	554	287	135	0	45	-11.9	-9.4	-12.1	-15.9	-9.4
16	30NiCrMo16	P1.B.+To	474	0	265	0	90	-18.6	-48.1	-6.7	-45.6	-16.9
17	30NiCrMo16	P1.B.+To	220	199	368	0	90	-4.5	-6.8	-4.1	-10	0
18	30NiCrMo16	P1.B.+To	470	299	261	0	90	-9.1	-37.4	-4.7	-42.5	-8.3
19	30NiCrMo16	P1.B.+To	527	287	129	0	90	-17.6	-20.5	-17.8	-27.1	-15.6
20	30NiCrMo16	P1.B.+To	433	472	240	0	90	2.8	-41.6	-11.8	-51.7	-11.3
21	30NiCrMo16	P1.B.+To	418	622	234	0	90	16.7	-38.5	-13.4	-54.8	-9.3
22	30NiCrMo16	P1.B.+To	0	299	396	0	0	-7.6	1	-4.8	-4.8	0.5
23	30NiCrMo16	P1.B.+To	0	486	411	0	0	-3.6	10	1	1	9.5
24	30NiCrMo16	P1.B.+To	0	655	364	0	0	-17	4.1	-9.9	-9.9	3.6
25	30NiCrMo16	P1.B.+To	482	0	268	0	0	-16.8	-2.7	-5.2	-5.2	-2.3
26	30NiCrMo16	P1.B.+To	207	299	350	0	0	-10.1	-1	-9.7	-9.7	-1.5
27	30NiCrMo16	P1.B.+To	474	294	265	0	0	-8.4	4.8	-3.6	-3.6	4.2
28	30NiCrMo16	P1.B.+To	584	281	142	0	0	-6.7	0	-6.3	-6.5	-0.7
29	30NiCrMo16	P1.B.+To	447	473	252	0	0	6.5	4.9	-7.4	-7.4	4.5
30	30NiCrMo16	P1.B.+To	425	635	223	0	0	17.9	1.9	-14.6	-14.6	1.4
31	C20	P1.B.+To	246	0	138	0	0	-1.2	6.9	6.9	4.1	13.4
32	C20	P1.B.+To	264	0	148	0	90	5.7	-23.1	1.9	-25.4	8.7
33	C20	P1.B.+To	246	0	138	0	45	-1.2	1.2	0	-4.1	6.9
34	EN-GJS800-2	P1.B.+To	199	0	147	0	0	-9.5	3	0.5	0	6.
35	EN-GJS800-2	P1.B.+To	245	0	142	0	90	0.8	-22	8.9	-19.6	5.3
36	EN-GJS800-2	P1.B.+To	228	0	132	0	0	-6.6	4.8	2.2	2.2	7.6
37	30NiCrMo16	Rot.B.+T	337	0	328	0	0	-8	-4.8	-8.3	-8.3	-5.3
38	30NiCrMo16	Rot.B.+T	482	0	234	0	0	-9.1	-8.9	-11.4	-11.4	-9.5

^a (Prop., proposal; DV., Dang Van; Pap., Papadopoulos; Cros., Crossland; Mo., Morel.

^b Bold REP values are those such that $|\text{REP}| \geq \pm 20\%$.

$$\text{REP}(\%) = \frac{\sigma_{a,\text{Exp}}^D - \sigma_{a,\text{Pred}}^D}{\sigma_{a,\text{Exp}}^D} \times 100 \quad (30)$$

In Table 2 we notice that the worst predictions are those of the local criteria of Dang Van and Crossland for non-proportional loadings and for tests with non-zero mean stress: up to a REP of 61%. The scatter of the REP for the Dang Van and Crossland criteria is the largest. All the REP values are illustrated in Fig. 14. Each bar of these histograms represents the number of test conditions for which the corresponding REP value is inside a 5% class width.

Fig. 14 shows that the REP histogram of our proposal is centered around -5% with a shorter scatter than other tested criteria. All the endurance limits predicted by our proposal are inside the smallest scatter band: only $\pm 20\%$. Furthermore, since all the other local criteria do not distinguish tension from bending, they are very sensitive to the parameter identification. When a combined loading is composed with plane bending, the criterion parameters (except for our proposal) were identified with the endurance limit in plane bending. *But, the use of the endurance limit in tension to identify material parameters of the local criteria, may generate more than 100% of REP.* This may be very dangerous for a design department.

5. Conclusion and prospects

After a quick overview of some fatigue criteria in which predictions are stress distribution dependent, a

volumetric energy based high-cycle multiaxial fatigue criterion is presented. The proposal is based on the damaging part of the volumetric density of the strain work given to the material per load cycle. This is an evolution of the energy based criterion proposed by Palin-Luc and Lasserre [3] which was limited to fully reversed sinusoidal loadings. The proposal uses a non-local approach based on the concept of volume influencing fatigue crack initiation. This volume is limited by an energy threshold W_g^* corresponding to the stress level σ^* below the usual endurance limit, above which micro-cracks can initiate in the material even if there is no macro-crack (detectable by stiffness decreasing of the specimen).

The predictions of this proposal depend on the stress-strain distribution inside the component; load types can thus be distinguished. This criterion is mean load sensitive; its predictions are in good agreement with many fatigue data from the literature. The phase shift effect on the endurance limit is also well predicted by the proposal. This effect is material dependent in biaxial tension. For a ductile material the effect is negative, but it is positive for brittle materials, as is reported in the literature. Under combined loadings, tension and torsion or bending and torsion, the criterion predictions are near the elliptic arc proposed by Gough and Pollard.

Since the volumetric stress-strain distribution is considered by the proposal, it will be interesting to test its accuracy with experiments on notched specimens. A post-processor for finite element analysis is needed to compute the volume influencing fatigue crack initiation if the component geometry is complex. We are now

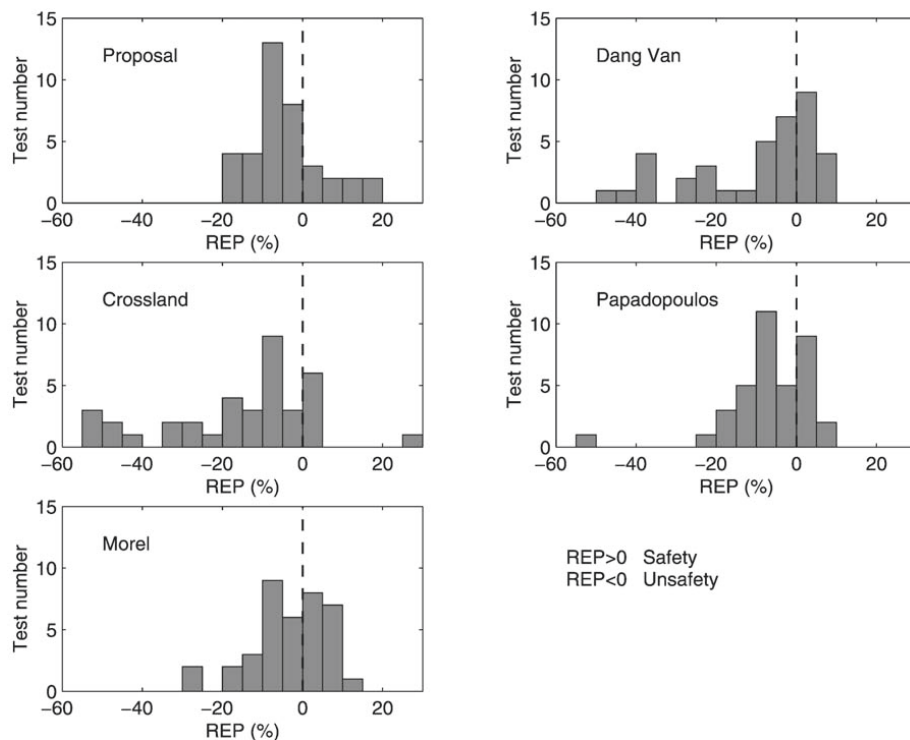


Fig. 14. Comparison of cumulative Relative Error of Prediction (%) class width = 5%.

working on this post-processor. The energy threshold W_g^* has to be correlated with micro-damage observations on different materials to understand the link between this threshold and the microstructure of the material. This proposal has also to be tested in middle cycle fatigue between 10^5 cycles and more. If its predictions are good, it should be possible to predict all the $S-N$ curves from a uniaxial one since the proposal is load type and mean load sensitive.

Acknowledgements

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Appendix: Brief description of the tested local criteria

The Crossland criterion is defined by a linear relationship between the maximum, on the loading period T , of the hydrostatic stress $\Sigma_{H,\max}$ and the amplitude τ_{a,J_2} of $\sqrt{J_2(t)}$, where $J_2(t)$ is the second invariant of the deviatoric stress tensor (31):

$$\tau_{a,J_2}(M) = \frac{1}{2\sqrt{2}} \max_{t' \in T} \left\{ \max_{t' \in T} \left[\sqrt{(\underline{S}(M,t) - \underline{S}(M,t')) : (\underline{S}(M,t) - \underline{S}(M,t'))} \right] \right\}. \quad (31)$$

$$\tau_{a,J_2}(M) + a \Sigma_{H,\max}(M) \leq b.$$

The material parameters a and b are given by the following equations: $a = (3\tau_{-1}^D / \sigma_{-1}^D) - \sqrt{3}$ and $b = \tau_{-1}^D$ where σ_{-1}^D is the fully reversed endurance limit for a uniaxial stress state (tension, plane bending or rotating bending) and τ_{-1}^D is the endurance limit in fully reversed torsion.

The Dang-Van model is a critical plane criterion based on a mesoscopic scale approach [37]. It can be expressed as Eq. (32) where $|\vec{\tau}_a(M, \vec{n}, t)|$ is the amplitude of the shear stress vector acting on the material plane orientated by the unit normal vector \vec{n} at the point M , $\Sigma_H(M)$ is the hydrostatic stress. The parameters c and d are identified from two experimental endurance limits, for instance: $c = (3\tau_{-1}^D / \sigma_{-1}^D) - (3/2)$ and $d = \tau_{-1}^D$.

$$\max_{\vec{n}} \left\{ \max_{t \in T} [\|\vec{\tau}_a(M, \vec{n}, t)\| + c \Sigma_H(M, t)] \right\} \leq d. \quad (32)$$

Depending on the value of the ratio $\tau_{-1}^D / \sigma_{\text{Ten},-1}^D$ of the material, Papadopoulos [38] proposed two high cycle multiaxial fatigue criteria based on a mesoscopic approach. These criteria are built with two root mean

square quantities. The first one, noted T_σ , (33), is related to the macroscopic resolved shear stress amplitude T_a acting on all the possible directions (located by the angle ψ) of a material plane (orientated by the unit normal vector \vec{n} , defined by the angles θ and ϕ). The second, noted M_σ , is the root mean square value of T_σ , on all the possible material planes (34).

$$T_\sigma(M, \theta, \phi) = \sqrt{\frac{1}{\pi} \int_{\psi=0}^{2\pi} T_a^2(\theta, \phi, \psi) d\psi}. \quad (33)$$

$$M_\sigma(M) = \sqrt{\frac{5}{8\pi^2} \int_{\phi=0}^{2\pi} \int_{\theta=0}^{\pi} T_\sigma^2(\theta, \phi) \sin\theta d\theta d\phi}. \quad (34)$$

Finally, the two endurance criteria are defined by Eqs (35) and (36). The material parameters are related, for instance, to the fully reversed endurance limits in torsion $\tau_{70,-1}^D$ and in uniaxial stress state σ_{-1}^D (tension, plane or rotating bending): $e = (3\tau_{-1}^D / \sigma_{-1}^D) - (3/2)$, $f = \tau_{-1}^D$ and $g = (3\tau_{-1}^D / \sigma_{-1}^D) - \sqrt{3}$, $h = \tau_{-1}^D$.

$$\text{If } 0.5 \leq \tau_{-1}^D / \sigma_{\text{Ten},-1}^D \leq 0.6,$$

$$\max_{\theta, \phi} [T_\sigma(M, \theta, \phi)] + e \Sigma_{H,\max}(M) \leq f. \quad (35)$$

$$\text{If } 0.6 \leq \tau_{-1}^D / \sigma_{\text{Ten},-1}^D \leq 0.8,$$

$$M_\sigma(M) + g \Sigma_{H,\max}(M) \leq h. \quad (36)$$

Morel [39,40] proposed a fatigue life calculation model based on the T_σ , Papadopoulos approach. The critical plane is the plane experiencing the maximum value of T_σ , noted $T_\Sigma = \max_{\theta, \phi} [T_\sigma(\theta, \phi)]$. According to this model, the number of cycles N_i for fatigue crack initiation at the point M is computed by the following equation for constant amplitude cyclic loadings:

$$N_i = p \ln \left(\frac{C_a}{C_a - T_s} \right) + q \left(\frac{\tau_{\text{lim}}}{C_a - \tau_{\text{lim}}} \right) - \frac{r}{C_a}. \quad (37)$$

where C_a is the maximum value of the resolved shear stress amplitude T_a on the critical plane. τ_{lim} is the limit value of the macroscopic shear stress amplitude on the critical plane (38). $\Sigma_{H,a}$ and $\Sigma_{H,m}$ are respectively the amplitude and the mean value of the hydrostatic stress, H is a phase shift coefficient: $H = T_\Sigma / C_a$, varying between $\sqrt{\pi}$ and $\sqrt{2\pi}$.

$$\tau_{\text{lim}} = \frac{-e \Sigma_{H,m} + f}{e \frac{\Sigma_{H,a}}{C_a} + H}. \quad (38)$$

This author assumes that a crystal of a metallic material has a mechanical behaviour which can be described by three phases: hardening, saturation and softening. The parameters p , q and r are related to these phases, their identification is detailed in [21], [40]. Since the Morel approach is a life calculation model, its application was done by computing the endurance limit corresponding to the given long life of the reference experimental data (10^6 cycles or more).

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volumetric energy based high cycle multiaxial fatigue criterio

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ERRATA

1 Triaxiality degree definition

In the published paper in International Journal of Fatigue (2003) Equation (13) is:

$$dT(M) = \frac{W_g^s(M)}{W_g(M)} \text{ and } W_g^s(M) = \int_T \langle \sigma_{kk} \dot{\epsilon}_{kk} \rangle dt$$

In fact, the GOOD equation is

$$dT(M) = \frac{W_g^s(M)}{W_g(M)} \text{ and } W_g^s(M) = \frac{1}{3} \int_T \left\langle \left(\sum_j \sigma_{jj} \right) \left(\sum_k \dot{\epsilon}_{kk} \right) \right\rangle dt$$

Nota: only the second term (after “and”) of equation (13) must be corrected.

2 Fatigue data

In Table 2 there is an error in the endurance limits of fatigue test No. 4.

The correct values are $\sigma_a^D = 558$ MPa (instead of 575) and $\bar{\sigma}^D = 428$ MPa (instead of 375) with: $\tau_a^D = \bar{\tau}^D = 0$ MPa and $\phi = 0$ degree; the REP values are correct.